

IN THE SPECIFICATION:

Please replace paragraph [0014] with the following amended paragraph [0014]:

[0014] Fig. 1 shows an excerpt of a cross section of the fuselage from a passenger plane, restricted to a sectional area of an aircraft passenger cabin 1. This arrangement would likely be familiar an expert in aircraft construction, and discloses relations from which the observer may recognize that a combustible interior cladding 3 is situated very close (in terms of the fuselage) to the external skin 2, which when installed together with the exterior skin 2 [[33]] encompasses a space 4 [[19]] within which the fuselage insulation (not shown on Fig. 1) is installed. If an exterior skin 2 traditionally realized with an aluminum material or aluminum alloy is installed in this configuration, the observer will be able to weigh the extent of a catastrophic fire of the kind described at the outset. The additional parts and elements of the interior equipment and fuselage structure shown on Fig. 1 and integrated into an aircraft passenger cabin 1 [[21]] (based on the example therein) will not be described, since they are considered to be irrelevant with respect to the present invention.

Please replace paragraph [0019] with the following amended paragraph [0019]:

[0019] The desired arrangement is coated by a resin layer or imbedded in a resin. The obtained exterior skin product with this material combination has a sandwich design. This sandwich design is adhesively bonded with a composite material and the mentioned metal material in layers (films), which yields a burn-through resistant behavior of the exterior skin relative to long-term exposure to flames from a fire. In addition, the sandwich design can be realized using a GLARE [[glare]] material, whose burn-through behavior is high.